



A NOVEL CONCEPT: TOWARDS SOLVING THE STARSHOT NANOCRAFT PROPULSION PROBLEM

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KeyWords

Concentric diamond beam, exoplanet, intergalactic sails, nanocraft, beamlets, beam arm, clearing beam, ultra-high-power, far field, beam radius, ring radius.

ABSTRACT

This paper presents a concept on ultra-high-power scaling capable of producing adequate laser beam power as the wind for propelling nanocrafts for intergalactic sails. The Concentric Diamond Beam (CDB) concept was conceptualized and developed with focus on the power requirement of the maiden starshot nanocraft propulsion project. In its simplest form, the concept involves beamlet sources arranged in concentric circles with optimal fill factor to ensure constructive interference at the far field. The concept consists of 33 equations derived to predict real world values of concept parameters required for early stage engineering and programming. Among others, the method makes it possible to determine the optimal number of beam arms for each ring, the distance covered by the nano-spacecraft, the angles at which beams from each ring should be shot to converge and the minimum distance of the clearing beam from the nanocraft. The clearing beam precedes the pushing beam to sweep the pathway for smooth sail. With eight underpinning assumptions, the concept aims at developing a novel approach that can generate adequate wind of propulsion to push nanocrafts to Prima Centauri b and other exoplanets in nearby gal-axies. The concept will also be useful as a design for the construction of ultra-high-power laser missile defense systems.

I. Background

The dream of sailing space with high-power laser propelled nanocrafts is age long with Robert Forward's paper - Round Trip Interstellar Travel Using Laser-Pushed Lightsail - marking a major milestone (feasibility study)¹. According to Robert Fugate (2017) the current generation is prepared for the implementation phase with some outlined physical, logistical, engineering, and legal challenges yet to be addressed².

The most popular among the physical problems is the struggle to gather adequate wind of propulsion to sail, direct and stabilize nanocrafts engineered for interstellar navigations^{3,4,5}. The roadmap is led by conceptual attempts of power scaling of laser beams to the order of 100 gigawatts. So far, Tandem pumping and beam combining have been the two methods developed by scientists over the years with the latter leading the discourse due to its outstanding advantages over the former^{6,7,8,9,10, 11,12}.

The arrival of Coherent and Incoherent beam combining techniques pushed the frontiers of practical laser power scaling to the current limit^{13,14,15,16,17} - the order of 100kW - which is below the required threshold for effective propulsion of nanocrafts. In order to realize the goal of laser beam propulsion and to benefit other applicable disciplines, there must be the possibility of efficiently combining millions of beams to obtain a single high-power beam output without compromising beam quality and stability^{17,18,19,20}. Thermal challenges, optical issues, nonlinear effects and pumping power challenges are the limiting factors to generating high power output especially for single mode fiber lasers^{22,23,24,25,26}. The five-year target set by Breakthrough Initiatives in 2017 to complete the conceptualization phase²⁷ is yet to be met seven years down the line.

This concept draws on heavily reviewed literature, related physical and mathematical theories to develop a novel approach of generating adequate laser power to push the first ever nanocraft to an exoplanet in Proxima Centauri star system and to future galaxies. This will brighten the chances of the human race to investigate Proxima Centauri b (a likely habitable planet in the Proxima Centauri system)³ and will broaden the scope of space science going forward. The concept will find its usefulness in laser missile weaponry technology and in building anti-missile defense systems

II. The Concentric Diamond Beam Concept

The concentric diamond Beam (CDR) concept consists of an arrangement of a finite number of laser beam sources in concentric rings to constitute a diamond-like structure as shown in Figure 1. As the beam angles are increased slowly and continuously, beam arms are lifted gradually, and P_i (point of convergence of all beams) is observed as a point object moving vertically upward from point P_0 to point P_3 and beyond. In the special scenario depicted in Figure 1, the beams from one ring initially converged at P_1 and then simultaneously increased beam angles gradually and continuously until they reached point P_2 and finally to point P_3 . Each ring consists of closely packed beams as discussed further in later part of this paper. There can be as many rings as possible depending on the power requirement of the task ahead. To visualize the CDR concept, imagine each point on the circumference of a circular disc as a source of laser beam positioned to converge at a common focus P_i , which is equidistant from all points on the circumference. For simplicity sake, if we further imagine just four points on the circumference contributing to the output beam such that the four points form the vertices of a square, then we will obtain a diamond-like structure like Figure 1.

The Concentric Diamond Beam approach dwells on the following assumptions:

1. All input beams are homogenous with the same optical and geometric characteristics.
2. The input beams coherently combine at the far field.
3. At the point of convergence, the beams are coherent hence they will reinforce one another to scale up output beam power.
4. As the beam exit the earth's gravitational field, the workdone against gravity approaches zero and less beam power will be required for further propulsion.
5. As the nanocraft approaches the distant planet (eg. Proxima Centauri b), gravitational pull by the planet further reduces workdone by the beam.
6. The minimum allowable increase in ring radius between two successive rings for maximum fill factor is $D = 2r$. Where D and r are beam diameter and radius, respectively.
7. Interstellar masses are asymmetric in shape to the clearing beam and will proceed in a random direction rather than the direction of sail when pushed by clearing beam.
8. The minimum effective value of beam angle occurs at $\alpha = 60^\circ$, at which Figure g becomes equilateral triangle as shown below.

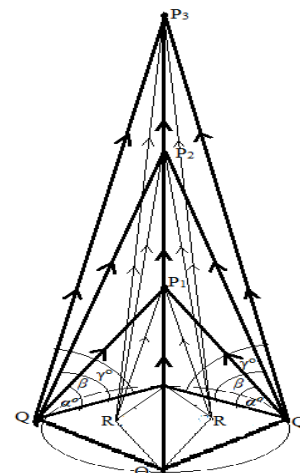


Figure 1. Beamlets from four vertices of a square touching the circumference of the first ring.

III. Inner Ring Arms and Outer Ring Arms

The diamond beam setup is made up of one inner ring of many closely packed beam arms surrounded by several outer rings. All the beam arms are oriented to focus at a point P_i , at a distance y_i , from the ground. Figure x shows the shooting of twelve beam arms of a single ring from the ground with O as the midpoint. All beam arms are equally spaced to ensure stability of the nanocraft motion.

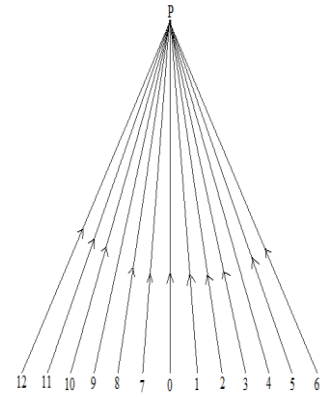


Figure x. Convergence of beams

Figure a. shows the arrangement of laser beam arms in the inner ring. If the radius of the ring formed by the densely packed laser beams is R and the radius of each beam in the ring is r , then from the figure b below,

$$\sin \alpha^o = \frac{r}{r+R}$$

$$\alpha^o = \sin^{-1}\left(\frac{r}{r+R}\right) \text{ ----- (a)}$$

From Figure a, the angle subtended by each circle is 2α at the center of the ring.

If there are n densely packed beam arms in the ring, then for the inner ring;

$$n_o \times 2\alpha^o = 360^o$$

$$n_o = \frac{\pi}{\sin^{-1}\left(\frac{r}{r+R}\right)} \text{ ----- (1)}$$

Equation (1) implies that, there can be finite number of laser beams in the inner ring. To get maximum value of n , $\sin^{-1}\left(\frac{r}{r+R}\right)$ must be minimum which occurs at -1, but this can not be possible as it will yield $n = -180$ (negative number of beams). The possible range of values of $\sin^{-1}\left(\frac{r}{r+R}\right)$ for which n is defined is $\{0 < \sin^{-1}\left(\frac{r}{r+R}\right) < 1\}$. Hence, for maximum n , the value of $\sin^{-1}\left(\frac{r}{r+R}\right)$ must be as close as possible to zero but not zero.

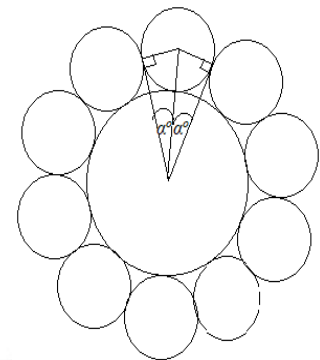


Figure a. Beam sources arranged in circular ring.

Demonstrating the above with hypothetical values (i.e beam diameter of 0.1cm and ring diameter of $R_0 = 100$ cm);

$$n_o = \frac{180}{\sin^{-1}\left(\frac{0.001}{0.001+1}\right)}$$

$n_o = 314162$ beam arms plus one middle beam.

If each beam arm is to contribute input beamlet of power 100kW, then a combined beam power of around 31GW will be produced by the inner ring. This is less than the estimated 50 GW threshold for the nanocraft propulsion mission. However, an addition of the first order ring will increase the output power to a value within the estimated range (50 GW – 100 GW) for nanocraft propulsion. If we were to maintain r at 0.1cm and reduce R_0 to as low as 5cm,

$$\text{then } n_o = \frac{180}{\sin^{-1}\left(\frac{0.001}{0.001+0.05}\right)}$$

$n_o = 160$ beam arms plus one middle beam.

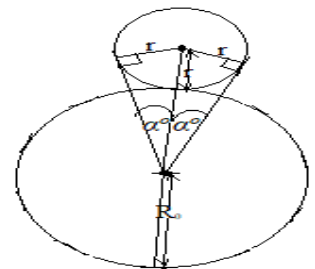


Figure b. Single beam held in position for simplicity of analysis

IV. Inner Ring Arms and Outer Ring Arms

The CDR setup has an inner ring surrounded by outer beam rings in a concentric fashion. The inner ring houses a single beam arm in its middle that fires directly at the target at 90^o to the horizontal. The first order ring envelopes the inner ring and the second, third and subsequent high rings follow in ascending order.

The first order ring has an incremental increase in radius of Γ . The new ring diameter becomes $R_0 + \Gamma$. From Figure c, the minimum value of Γ is $D = 2r$. From equation (1);

$$n_1 = \frac{\pi}{\sin^{-1}\left(\frac{r}{r+R_0}\right)} \quad \text{but } R_1 = R_0 + \Gamma$$

$$n_1 = \frac{\pi}{\sin^{-1}\left(\frac{r}{r+\Gamma+R_0}\right)} \quad \text{The minimum allowable value of } \Gamma \text{ will be } \Gamma_0 = D = 2r$$

$$n_1 = \frac{\pi}{\sin^{-1}\left(\frac{r}{r+2r+R_0}\right)}$$

$$n_1 = \frac{\pi}{\sin^{-1}\left(\frac{r}{3r+R_0}\right)} \quad \text{----- (2)}$$

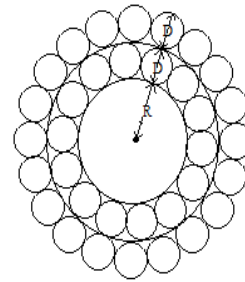


Figure c. First order ring

Comparing equation (1) and equation (2), it can be inferred that, for the same value of r and R , $n_1 < n_0$. For instance, subjecting equation (2) to the values in our earlier scenario ($r = 0.1$ cm and $R = 5$ cm) reveals that, $n_1 \approx n_0 + 6$.

The second order ring is the next ring of laser beam arms to the first order ring. Just as the first order ring, the ring radius increases by an incremental value of Γ . The ring radius of the second order ring becomes $R_2 = R_1 + \Gamma = R_0 + 2\Gamma$

Again, from equation (1),

$$n_2 = \frac{\pi}{\sin^{-1}\left(\frac{r}{r+R_2}\right)} \quad \text{but } R_2 = R_0 + 2\Gamma$$

For minimum value of Γ , $\Gamma = D = 2r$

$$R_2 = R_0 + 4r$$

$$n_2 = \frac{\pi}{\sin^{-1}\left(\frac{r}{r+4r+R_0}\right)}$$

$$n_2 = \frac{\pi}{\sin^{-1}\left(\frac{r}{5r+R_0}\right)} \quad \text{----- (3)}$$

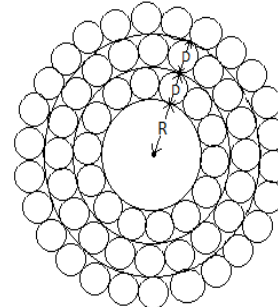


Figure e. Beams in the 2nd order ring.

Again, iterating with $0.1 \text{ cm} \leq R \leq 100 \text{ cm}$ and constant $r = 0.1$ cm, reveals that $n_1 \approx n_0 + 6$ and $n_2 \approx n_0 + 2(6)$

The third order ring comes after the second order ring and has an extension of ring radius by Γ . This implies that, $R_3 = R_2 + \Gamma$ From equation (1),

$$n_3 = \frac{\pi}{\sin^{-1}\left(\frac{r}{r+R_3}\right)} \quad \text{but } \Gamma = D = 2r$$

$$n_3 = \frac{\pi}{\sin^{-1}\left(\frac{r}{r+\Gamma+R_2}\right)}, \quad \text{but } R_2 = R_0 + 2\Gamma$$

$$n_3 = \frac{\pi}{\sin^{-1}\left(\frac{r}{r+3\Gamma+R_0}\right)}$$

$$n_3 = \frac{\pi}{\sin^{-1}\left(\frac{r}{r+6r+R_0}\right)}$$

$$n_3 = \frac{\pi}{\sin^{-1}\left(\frac{r}{7r+R_0}\right)} \quad \text{----- (4)}$$

We have shown from iteration with values of R between 0.1 cm and 100 cm that,

$$n_0 \approx n_0 + 0(6), \quad n_1 \approx n_0 + 1(6), \quad n_2 \approx n_0 + 2(6), \quad n_3 \approx n_0 + 3(6)$$

$$\text{Hence, } n_i \approx n_0 + i(6) \quad \text{----- (5)}$$

for $i = 0, 1, 2, 3, \dots, N$.

From iteration with the same values of R and r , we can generalize equation (1) as

$$n_i = \frac{\pi}{\sin^{-1}\left(\frac{r}{(2i+1)r+R_0}\right)} \quad \text{----- (6)}$$

Also, we can show that,

$$R_i = R_0 + i\Gamma \quad \text{----- (7)}$$

for $i = 0, 1, 2, 3, \dots, N$

and for all values of R at a beam radius of 0.1 cm.

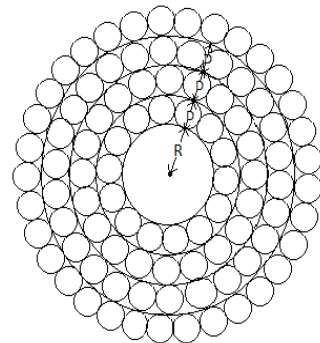


Figure f. Beams in the 3rd order ring.

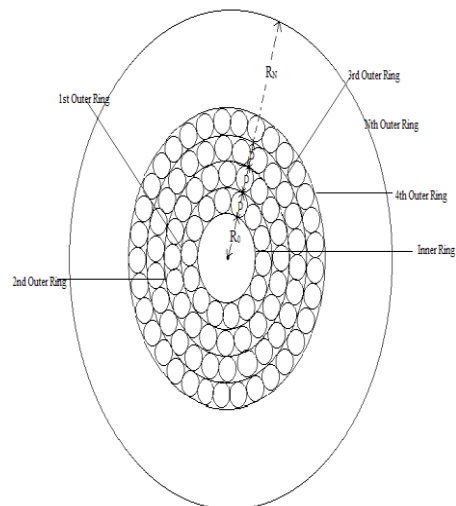


Figure g. Beams in the Nth ring.

V. Sail Displacement and Beam Displacement

For simplicity sake, we consider a single beam arm targeted at a fix point P in space as shown on Figure j. This can be extracted from Figure 1. Considering the cosine rule,

$$y^2 = x^2 + z^2 - 2xz \cos \alpha \quad \text{----- (a)}$$

but z can not be measured directly and so with sine rule, we attempt to eliminate it.

$$\frac{y}{\sin \alpha} = \frac{z}{\sin \theta}$$

$$z = \frac{\sin \theta}{\sin \alpha} y$$

Substituting this into equation (a),

$$y^2 = x^2 + \left(\frac{\sin \theta}{\sin \alpha} y\right)^2 - 2x\left(\frac{\sin \theta}{\sin \alpha} y\right) \cos \alpha$$

$$y^2 = x^2 + \frac{\sin^2 \theta}{\sin^2 \alpha} y^2 - 2xy \frac{\sin \theta \cos \alpha}{\sin \alpha}$$

rearranging the above equation,

$$\left[1 - \frac{\sin^2 \theta}{\sin^2 \alpha}\right] y^2 + \left[2x \frac{\sin \theta \cos \alpha}{\sin \alpha}\right] y - x^2 = 0$$

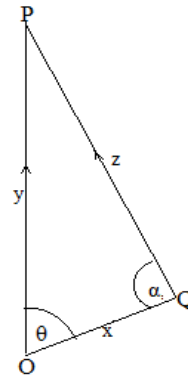


Figure j. Single beam focused at point P.

For simplicity sake, we consider a single beam arm targeted at a fix point P in space as shown on Figure j. This can be extracted from Figure 1. Considering the cosine rule,

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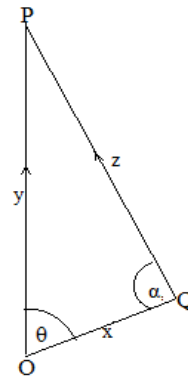


Figure j. Single beam focused at point P.

Comparing this to the general equation, $ay^2 + by + c = 0$ with the solution $y = \frac{-b \pm \sqrt{b^2 - 4ac}}{2a}$,

$$a = 1 - \frac{\sin^2 \theta}{\sin^2 \alpha} \quad b = 2x \frac{\sin \theta \cos \alpha}{\sin \alpha} \quad \text{and} \quad c = -x^2$$

$$y = \frac{-2x \frac{\sin \theta \cos \alpha}{\sin \alpha} \pm \sqrt{\left(2x \frac{\sin \theta \cos \alpha}{\sin \alpha}\right)^2 - 4\left(1 - \frac{\sin^2 \theta}{\sin^2 \alpha}\right)(-x^2)}}{2\left(1 - \frac{\sin^2 \theta}{\sin^2 \alpha}\right)}$$

$$y = \frac{-2x \frac{\sin \theta \cos \alpha}{\sin \alpha} \pm 2x \sqrt{\left(\frac{\sin^2 \theta \cos^2 \alpha}{\sin^2 \alpha}\right) + \left(1 - \frac{\sin^2 \theta}{\sin^2 \alpha}\right)}}{2\left(1 - \frac{\sin^2 \theta}{\sin^2 \alpha}\right)}$$

$$y = \frac{-2x \frac{\sin \theta \cos \alpha}{\sin \alpha} \pm 2x \sqrt{\left(\frac{\sin^2 \theta \cos^2 \alpha}{\sin^2 \alpha}\right) + \left(1 - \frac{\sin^2 \theta}{\sin^2 \alpha}\right)}}{2\left(1 - \frac{\sin^2 \theta}{\sin^2 \alpha}\right)}$$

$$y = \frac{-2x \left[\frac{\sin \theta \cos \alpha}{\sin \alpha} \pm \sqrt{\left(\frac{\sin^2 \theta \cos^2 \alpha}{\sin^2 \alpha}\right) + \left(1 - \frac{\sin^2 \theta}{\sin^2 \alpha}\right)} \right]}{2\left(1 - \frac{\sin^2 \theta}{\sin^2 \alpha}\right)}$$

$$= \frac{-\left[\frac{\sin\theta \cos\alpha}{\sin\alpha} \pm \sqrt{\left(\frac{\sin^2\theta \cos^2\alpha}{\sin^2\alpha}\right) + \left(1 - \frac{\sin^2\theta}{\sin^2\alpha}\right)}\right]x}{\left(1 - \frac{\sin^2\theta}{\sin^2\alpha}\right)}$$

$$y = \frac{-\left[\frac{\sin\theta \cos\alpha}{\sin\alpha} \pm \sqrt{\left(\frac{\sin^2\theta \cos^2\alpha}{\sin^2\alpha}\right) + \left(1 - \frac{\sin^2\theta}{\sin^2\alpha}\right)}\right]x}{\left(1 - \frac{\sin^2\theta}{\sin^2\alpha}\right)}$$

$$y = \frac{-\left[\frac{\sin\theta \cos\alpha}{\sin\alpha} \pm \sqrt{1 - \sin^2\theta}\right]x}{\left(1 - \frac{\sin^2\theta}{\sin^2\alpha}\right)}$$

$$y = \frac{-\left[\frac{\sin\theta \cos\alpha}{\sin\alpha} \pm \sqrt{\cos^2\theta}\right]x}{\left(1 - \frac{\sin^2\theta}{\sin^2\alpha}\right)}$$

$$y = \frac{-\left[\frac{\sin\theta \cos\alpha}{\sin\alpha} \pm \cos\theta\right]x}{\left(1 - \frac{\sin^2\theta}{\sin^2\alpha}\right)}$$

$$y = \frac{-\left[\frac{\sin\theta \cos\alpha \pm \sin\alpha \cos\theta}{\sin\alpha}\right]x}{\frac{\sin^2\alpha - \sin^2\theta}{\sin^2\alpha}}$$

$$y = \frac{\sin\theta \sin\alpha \cos\alpha \pm \sin^2\alpha \cos\theta}{\sin^2\theta - \sin^2\alpha} * x \quad \text{----- (8)}$$

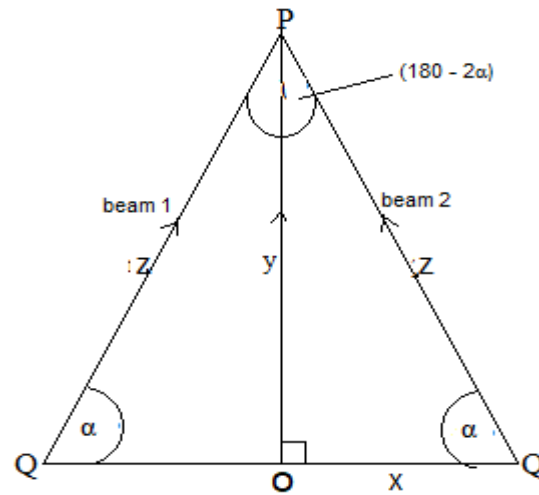


Figure k. Two beams from inner ring converging at P.

where y is the sail distance and alpha is the beam angle. If we maintain theta at 90°, then;

$$y = \frac{\sin\alpha \cos\alpha}{\cos^2\alpha} * x$$

$$y = \frac{\sin\alpha}{\cos\alpha} * x$$

$$y = x \tan\alpha \quad \text{----- (9)}$$

From equation (9), y depends on x and tan alpha. But x is finite because we cannot have infinite ground space available. This implies that alpha lies between 0° and 90°. i.e. 0° < alpha < 90°. As alpha approaches 90°, y approaches infinity which is desirable. With such a setup, the concentric beams can continue to converge/combine to push the target to the desirable destination.

From Figure j, if we fix theta at 90° and complete the other half of the triangle, we obtain Figure k. Applying Pythagoras theorem on this, z² = y² + x² and substituting y from equation (9),

$$z^2 = (x \tan\alpha)^2 + x^2$$

$$z = \sqrt{[x^2(\tan\alpha)^2 + 1]}$$

$$z = x\sqrt{(\tan^2\alpha + 1)} \quad \text{----- (10)}$$

where z is the total distance travelled by the laser beam known as beam flight.

This equation is useful as it helps to determine how far the nanocraft has travelled based on the current beam angle and the distance of the beam source from the middle beam O. This is also useful in determining the beam velocity i.e. Beam velocity $V_B = \frac{\Delta Z}{\Delta t}$.

$$V_B(t) = \frac{dZ}{dt} = \frac{d(x\sqrt{(\tan^2\alpha + 1)})}{dt} \quad \text{----- (11)}$$

$$\text{and the beam acceleration will follow as } a_B(t) = \frac{dZ}{dt} = \frac{d(x\sqrt{(\tan^2\alpha + 1)})}{dt} \quad \text{----- (12)}$$

VI. Beam Shooting Angle

It is the angle at which all beam arms in a given ring must be shot to converge at a common focus, equidistant from all beams in the ring.

From Figure k and applying equation (8),

$$\alpha = \tan^{-1}\left(\frac{y}{x}\right) \quad \text{----- (13)}$$

If we bring on first order ring to Figure k, we obtain Figure 3 as shown.

Applying equation (9), $y = x_1 \tan\beta$

$y = (x_0 + \Gamma) \tan\beta$ since $x_1 = x_0 + \Gamma$

but for maximum n, $\Gamma = D = 2r$

$y = (x_0 + 2r) \tan\beta$

$\tan\beta = \frac{y}{(2r+x)}$

$\beta = \tan^{-1}\left(\frac{y}{(2r+x_0)}\right) \quad \text{----- (14)}$

For the first order ring to converge with the inner ring, ring arms must be oriented at beta° to the horizontal. If we add second order ring to Figure 3, we obtain Figure 4 as shown below.

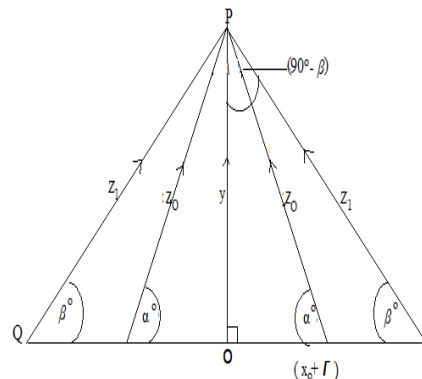


Figure 3. shooting angles of beams in the inner ring and 1st order ring.

It follows from equation (9) that, $y = (x_0 + 2r) \tan \gamma^0$

Again for maximum n, $r = D = 2r$

$$y = (x_0 + 4r) \tan \gamma^0$$

$$\tan \gamma^0 = \left(\frac{y}{x_0 + 4r} \right)$$

$$\gamma^0 = \tan^{-1} \left(\frac{y}{x_0 + 4r} \right) \text{ ----- (15)}$$

In a similar way, if we impose the third order ring on Figure 4, we obtain Figure 5 below. As more and more orders are added, Figure 5 begins to assume the shape of Figure x discussed earlier.

From Figure 5. It follows from equation (9) that, $y = (x_0 + 3r) \tan \eta^0$. Inserting $r = D = 2r$ into the above,

$$y = (x_0 + 6r) \tan \eta^0$$

$$\tan \eta^0 = \frac{y}{(x_0 + 6r)}$$

$$\eta^0 = \tan^{-1} \left(\frac{y}{(x_0 + 6r)} \right) \text{ ----- (16)}$$

If we replace α, β, γ and η with $\alpha_0, \alpha_1, \alpha_2$ and α_3 respectively. Comparing equations (13), (14), (15) and (16) we form the generalization that;

$$\alpha_i = \tan^{-1} \left(\frac{y}{(x_0 + 2ir)} \right) \text{ ----- (17)}$$

For $i = 0, 1, 2, 3, \dots, N$

Where α_i is the beam shooting angle for inner ring beams, first order beams, second order beams, third order beams and so on.

It follows from equation (9) that, $y = (x_0 + 3r) \tan \eta^0$

Inserting $r = D = 2r$ into the above,

$$y = (x_0 + 6r) \tan \eta^0$$

$$\tan \eta^0 = \frac{y}{(x_0 + 6r)}$$

$$\eta^0 = \tan^{-1} \left(\frac{y}{(x_0 + 6r)} \right) \text{ ----- (16)}$$

If we consider α, β, γ and η as the angles subtended by the same beam arm at different positions, then we can replace α, β, γ and η with $\alpha_0, \alpha_1, \alpha_2$ and α_3 respectively. Comparing equations (13), (14), (15) and (16) we can form a generalization that;

$$\alpha_i = \tan^{-1} \left(\frac{y}{(x_0 + 2ir)} \right) \text{ ----- (17)}$$

For $i = 0, 1, 2, 3, \dots, N$

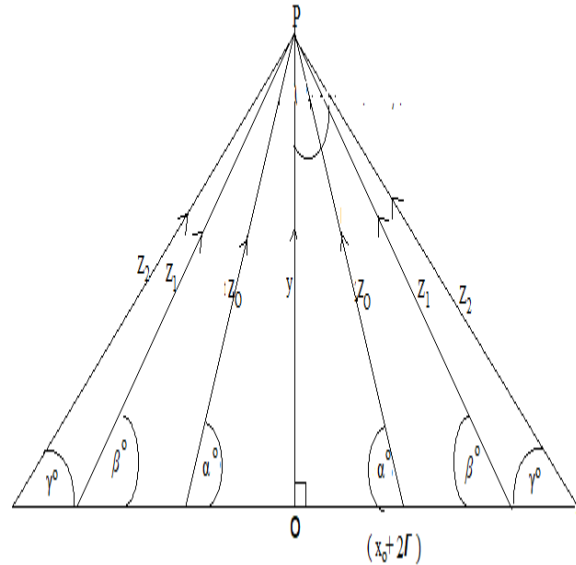


Figure 4. Shooting angles of beams in the inner ring, 1st and 2nd order rings.

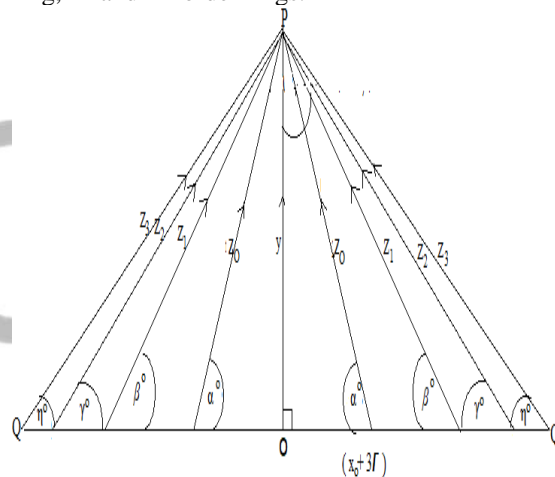


Figure 5. Shooting angles of the inner ring, 1st, 2nd, and 3rd order rings.

VII. Beam Velocity and Path Difference

From Figure k, $z_0^2 = y^2 + x_0^2$

But $y = x_0 \tan \alpha_0$

$$z_0^2 = (x_0 \tan \alpha_0)^2 + x_0^2$$

$$z_0^2 = x_0^2 \tan^2 \alpha_0 + x_0^2$$

$$z_0 = \sqrt{x_0^2 \tan^2 \alpha_0 + x_0^2} \text{ ----- (18) or } z_0 = x_0 \sqrt{\tan^2 \alpha_0 + 1}$$

$$z_0 = x_0 \sqrt{\sec^2 \alpha_0}, \quad z_0 = \frac{x_0}{\cos \alpha_0} \text{ ----- (19)}$$

Also, from Figure 3

$$z_1^2 = y^2 + x_1^2, \quad \text{and } x_0 = x_0 + r = x_0 + 2r$$

Substituting this into the above,

$$z_1^2 = y^2 + (x_0 + 2r)^2 \quad \text{but } y = x_0 \tan \alpha_0$$

$$z_1^2 = x_0^2 \tan^2 \alpha_0 + x_0^2 + 4x_0r + 4r^2$$

$$z_1 = \sqrt{x_0^2 \tan^2 \alpha_0 + x_0^2 + 4x_0r + 4r^2} \text{ ----- (20)}$$

Equation (20) gives the beam distance of the first order beams in terms of x_0, α_0 and r . Also, from Figure 3,

$$\sin \alpha_0 = \frac{y}{z_0} \text{ ----- (a)}$$

$$\sin\alpha_1 = \frac{y}{z_1} \quad \text{----- (b)}$$

Dividing equation (b) by equation (a), $\frac{\sin\alpha_1}{\sin\alpha_0} = \frac{z_1}{z_0}$

$$z_1 = \frac{\sin\alpha_1}{\sin\alpha_0} z_0 \quad \text{----- (21), Taking the derivative of both sides of equation (21),}$$

$$\frac{dz_1}{dt} = \frac{\sin\alpha_1}{\sin\alpha_0} \frac{dz_0}{dt} \quad \text{and } V_1(t) = \frac{\sin\alpha_1}{\sin\alpha_0} V_0(t) \quad \text{----- (22)}$$

This implies that, for the beams in the two rings to arrive at point P at the same time, V_1 must be $(\frac{\sin\alpha_1}{\sin\alpha_0})$ times faster than V_0 , a condition difficult to achieve with lasers. Instead we focus on manipulating the time of trigger of beams for both rings. In the real world, we maintain the same velocity for all beams ($V_0 = V_1 = V$) and then calculate the respective times of arrival for the beams in different rings as follows;

$$t_0 = \frac{z_0}{v} \quad \text{and } t_1 = \frac{z_1}{v}$$

$$\text{The trigger time difference is then calculated as } \Delta t = t_1 - t_0 = \frac{z_1 - z_0}{v}$$

$(z_0 - z_1)$ is the path difference between beams in the first order ring and those in the inner ring. In effect, the setup must be programmed in such a way as to allow the inner beams to be triggered $(\frac{z_1 - z_0}{v})$ seconds earlier than beams in the first order ring. Since we can measure x_0 , α_0 and r from the setup, it is easier to predict z_1 and z_0 from equations (18) and (20). Alternatively, from Figure 3,

$$z_1^2 = y^2 + x_1^2 \quad \text{but } y = x_1 \tan\alpha_1$$

$$z_1^2 = x_1^2 \tan^2\alpha_1 + x_1^2$$

$$z_1 = \sqrt{[x_1^2 \sec^2\alpha_1]}$$

$$z_1 = \frac{x_1}{\cos\alpha_1} \quad \text{----- (23)}$$

Also, from Figure 4,

$$z_2^2 = y^2 + x_2^2 \quad \text{but } y = x_2 \tan\alpha_2$$

$$z_2^2 = x_2^2 \tan^2\alpha_2 + x_2^2$$

$$z_2^2 = x_2^2 \sec^2\alpha_2$$

$$z_2 = \frac{x_2}{\cos\alpha_2} \quad \text{----- (24)}$$

Also from Figure 4,

$$z_2^2 = y^2 + x_2^2$$

$$x_2 = x_0 + 2r = x_0 + 4r$$

Substituting this into the above,

$$z_2^2 = y^2 + (x_0 + 4r)^2 \quad \text{but } y = x_0 \tan\alpha_0$$

$$z_2^2 = x_0^2 \tan^2\alpha_0 + x_0^2 + 8x_0r + 16r^2$$

$$z_2 = \sqrt{x_0^2 \tan^2\alpha_0 + x_0^2 + 8x_0r + 16r^2} \quad \text{----- (25)}$$

From Figure 5,

$$z_3^2 = y^2 + x_3^2 \quad \text{but } y = x_3 \tan\alpha_3$$

$$z_3^2 = x_3^2 \tan^2\alpha_3 + x_3^2$$

$$z_3^2 = x_3^2 (\tan^2\alpha_3 + 1)$$

$$z_3 = \frac{x_3}{\cos\alpha_3} \quad \text{----- (26)}$$

$$\text{Alternatively, } z_3^2 = y^2 + x_3^2$$

$$x_3 = x_0 + 3r = x_0 + 6r$$

Substituting this into the above,

$$z_3^2 = y^2 + (x_0 + 6r)^2 \quad \text{but } y = x_0 \tan\alpha_0$$

$$z_3^2 = x_0^2 \tan^2\alpha_0 + x_0^2 + 12x_0r + 36r^2$$

$$z_3 = \sqrt{x_0^2 \tan^2\alpha_0 + x_0^2 + 12x_0r + 36r^2} \quad \text{----- (27)}$$

Comparing equations (18), (20), (25) and (27), we can state that,

$$z_i = \sqrt{x_0^2 (\tan^2\alpha_0 + 1) + 4ix_0r + 4i^2r^2}$$

$$z_i = \sqrt{x_0^2 \sec^2\alpha_0 + 4ir(x_0 + ir)} \quad \text{----- (28)}$$

where z_i is the beam displacement or beam path for the beams in rings $i = 0, 1, 2, 3, \dots, N$.

Similarly, comparing equations (19), (23), (24) and (26), we can form the generalization that,

$$z_i = \frac{x_i}{\cos\alpha_i} \quad \text{----- (29)}$$

VIII. Vertical Motion (Sail Displacement)

If all the beam arms in each ring are programmed to raise gradually by increasing the beam angles, then point P will be viewed as an object moving vertically upwards. If we keep x constant and choose to vary y and α , we obtain Figure 6 as depicted below. From the diagram, the angle between the two beams converging at P_0 is $(180^\circ - 2\alpha_0)$. At P_1, P_2 and P_3 the angles are $(180^\circ - 2\alpha_1), (180^\circ - 2\alpha_2)$ and $(180^\circ - 2\alpha_3)$ respectively.

In general, the angle subtended at the point of convergence by opposite beams in the i^{th} ring is $(180^\circ - 2\alpha_i)$.

From our previous discussions, it can be shown that; $y_i = x \tan \alpha_i$ ----- (30)

For a constant value of x

Equation (30) predicts the position of the beam, P_i from the ground for a given beam angle, α_i . This equation can also be useful in determining how far the nanocraft has travelled from earth.

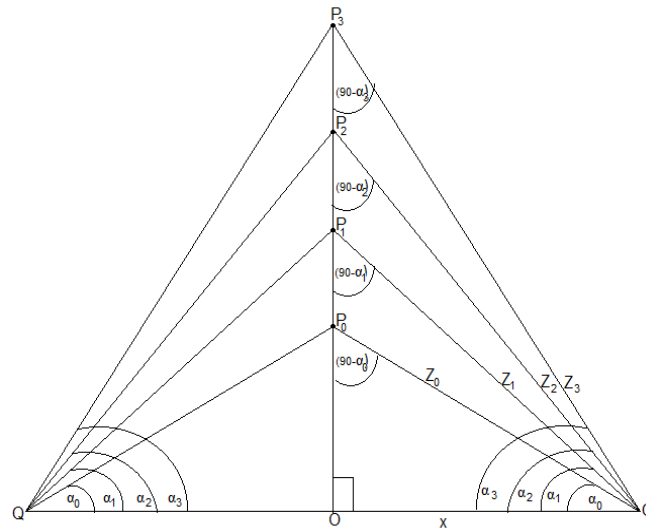


Figure 6. Vertical movement of two beams in the same ring form point P_1 to point P_2 .

IX. Path Clearing

One of the challenges of the nanocraft propulsion concept is the possibility of continuous collision between the spacecraft and interstellar masses leading to unwanted retardation or total abortion of mission³⁷. As long as the probability of occurrence of collision is greater than zero, it is necessary to address this challenge. From our analysis of vertical displacement, the point of convergence of beams in the ring can always be predicted. And so, in the last ring (clearing ring), all beams are programmed to converge at a certain distance ξ , ahead of the nanocraft. The clearing beam is expected to sweep the pathway of the spacecraft to reduce collisions. As shown in Figure 7, the minimum value of ξ will depend on the size of the spherical nanocraft.

Applying the Pythagoras theorem to $\Delta K_i O Q$,

$$(\xi + y_i)^2 = x^2 + S_i^2 \text{ ----- (n)}$$

Also, from sine rule with $\Delta K_i P_i Q$,

$$\frac{\xi}{\sin(\sigma_i - \alpha_i)} = \frac{z_i}{\sin(90^\circ - \sigma_i)}$$

$$\xi = \frac{\sin(\sigma_i - \alpha_i)}{\cos \sigma_i} z_i \text{ ----- (m)}$$

Again using the sine rule with $\Delta K_i O Q$,

$$\frac{\xi + y_i}{\sin \sigma_i} = \frac{x}{\sin(90^\circ - \sigma_i)}$$

$$\xi = \frac{\sin \sigma_i}{\cos \sigma_i} x - y_i \text{ ----- (o)}$$

Comparing equation (m) and equation (o),

$$\frac{\sin(\sigma_i - \alpha_i)}{\cos \sigma_i} z_i = \frac{\sin \sigma_i}{\cos \sigma_i} x - y_i$$

$$\left[\frac{\sin \sigma_i \cos \alpha_i}{\cos \sigma_i} - \frac{\sin \alpha_i \cos \sigma_i}{\cos \sigma_i} \right] * z_i = \frac{\sin \sigma_i}{\cos \sigma_i} x - y_i$$

$$z_i \tan \sigma_i \cos \alpha_i - z_i \sin \alpha_i = x \tan \alpha_i - y_i$$

$$\tan \sigma_i = \frac{z_i \sin \alpha_i - y_i}{z_i \cos \alpha_i - x}$$

$$\sigma_i = \tan^{-1} \left(\frac{z_i \sin \alpha_i - y_i}{z_i \cos \alpha_i - x} \right) \text{ ----- (31)}$$

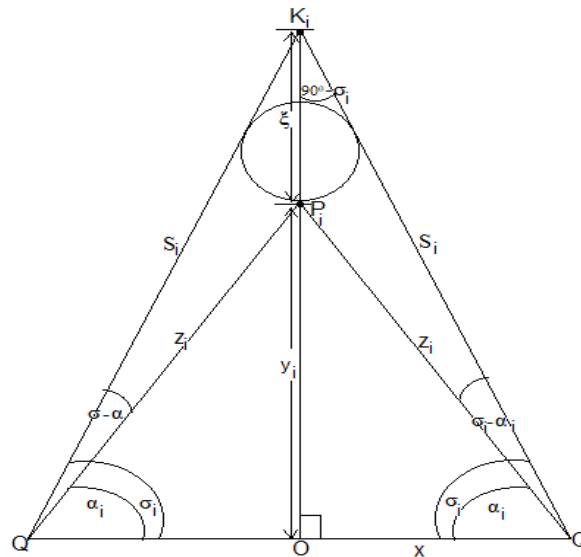


Figure 7. Path clearing beam ahead of the sail.

Since x is always known and y_i can be determined from $y_i = x \tan \alpha_i$, it is possible to predict the value of σ_i from equation (31). The value of z_i can be calculated from equation (28) or (29).

σ_i is known as the minimum allowable angle of the clearing beam. If the clearing beams are oriented at any angle less than σ_i , they will hit the nanocraft and the purpose for their creation will be defeated.

Substituting equation (31) into equation (o)

$$\xi = \frac{\sin \sigma_i}{\cos \sigma_i} x - y_i = x \tan \sigma_i - y_i$$

$$\xi = x \tan \left[\tan^{-1} \left(\frac{z_i \sin \alpha_i - y_i}{z_i \cos \alpha_i - x} \right) \right] - y_i$$

$$\xi = \frac{z_i \sin \alpha_i - y_i}{z_i \cos \alpha_i - x} x - y_i \text{ ----- (32)}$$

ξ is the minimum allowable distance between beams in the i^{th} ring and beams in the clearing ring.

Assumption:

Because interstellar masses are asymmetric in shape, the clearing beam will hit them off the pathway of the nanocraft as against sailing them ahead of the nanocraft perpetually.

Substituting equation (32) into equation (n),

$$\left[\left(\frac{z_i \sin \alpha_i - y_i}{z_i \cos \alpha_i - x} x - y_i \right) + y_i \right]^2 = x^2 + S_i^2$$

$$\left(\frac{z_i \sin \alpha_i - y_i}{z_i \cos \alpha_i - x} x \right)^2 - x^2 = S_i^2$$

$$S_i = \sqrt{\left[\left(\frac{z_i \sin \alpha_i - y_i}{z_i \cos \alpha_i - x} x \right)^2 - x^2 \right]} \quad \text{----- (33)}$$

X. Discussions

From the above, the paper showed that, laser beam power can be unlimitedly scaled by actively combining inputs from more beam arms. To achieve far field constructive interference of coherent beam inputs, the beams are packed to attain maximum fill factor⁴⁷. This is desirable as more beamlets also imply higher combined beam output power. Inferring from equations (5) and (6), each ring can contain a finite number of beam arms for which coherent inference is expected to take place at the far field. The number of possible beam arms in a ring depends both on the beam radius and the ring radius. It was revealed that, for a given value of beam radius, the inverse sine of the ratio of beam radius to the sum of beam radius and ring radius must be as close as possible to zero, in order to achieve optimal number of beam arms. This implies that, more iterations will be required in real project implementation or simulation experiments to ascertain the best ring diameter that will match available ground space and beam radius. The optimal incremental increase in ring radius is predetermined by equation (7). This suggests that, more and more high-order rings can be added until the power requirement for the given project is met.

The results showed that, beams in the same ring can be programmed to converge at the same spot or in a small circular shape in which case we have spot convergence and circular convergence, respectively. This suggests two separate trials in determining the most stable module for the maiden project. In either case, the sail distance is predictable by equation (8) or equation (9). From equation (9), it can be predicted that, with the range of angles available to beamlets in the rings, the beams can travel close to infinity depending on beam angle orientation. Zero and ninety are forbidden values of beam angle and serve as an important note for engineers and programmers during prototype construction.

Beam flight or beam displacement can be predicted with equation (10) which is useful in determining whether beams from different rings positioned to converge at a point will arrive at the same time or not. Following the classical analysis (without interest in the relativistic relations) of equation (9), difference in time of arrival between beams from different rings exist. To circumvent this challenge, time differences should be considered during programming to ensure that, beams from higher order rings take off (by a calculated factor) before beams in lower rings. This arrangement ensures that all beams arrive at the same time to result in power gains from constructive interference at the far field. To achieve this, beams in the various rings must be shot at specific angles (beam shooting angles) as predicted by equation (17). It follows from simple iteration that, for the same displacement covered by the sail, beam angle decreases as we move outwards from inner ring towards the Nth ring. This is desirable as it implies that beams from different rings can never cross paths before reaching the point of interference.

Beam velocity is predictable with equation (22) and this can be an indicator of how fast or slow the nanocraft is travelling. This will serve as useful guide in determining how many years we need to arrive at Prima Centauri at a given rate of acceleration. The extremely high expected beam velocity reveals that, momentum will be high and impact of collisions with small interstellar masses may be significant. Especially if collisions are continuous. The Path Clearing beam sweeps the pathway off celestial debris to allow free travel of the nanocraft. Equation (31) predicts the minimum angle at which the path clearing beam can be shot without touching a spherical sail, while equation (32) predicts the minimum distance from the nanocraft at which path clearing beam can be targeted without touching the sail. Both equations forbid a path clearing beam angle of ninety degrees. Finally, equation (33) makes it possible to estimate the distance travelled by the path clearing beam. This distance may be used in varied applications including determining when to switch off path clearing beam to ensure energy efficiency.

XI. Conclusion

The CDB concept makes it possible to scale up laser beam power towards infinitesimally large magnitudes for intergalactic sails. This can be possible by completely avoiding passive components in the setup. Aside the ultrahigh power achievement, the concept allows for improved beam quality as constructive interference will take place at the far field. The combined power output from the arrangement depends on the power levels of the individual beamlets and especially the number of beam arms present. By extension, the number of rings also affect the magnitude of output power. To ensure constructive interference at the far field, lower order rings must be fully filled with beam arms before adding higher order rings.

Convergence of beams in a ring can be circular or spot and in both approaches the position of output beam depends on beam angles. Beamlets can be oriented to converge at positions close to infinity depending on project requirements and by adjusting beam angles. Beams from different rings will arrive at the same spot at different times and so trigger time for the various rings must be

manipulated to achieve simultaneous arrival of all beams. With the CDR concept, no two beamlets can cross paths before reaching the point of convergence. The concept makes it possible to predict the velocity of sail and by extension the duration of the entire journey at a given velocity and acceleration. The concept also allow for path clearing to avoid the consequence of continuous collision with celestial debris.

Overall, the Starshot Nanocraft Propulsion mission should no longer be held backward by the physical limitation of low power scaling options. This paper presents an option to high-power scaling of laser input beams beyond the predicted range for propelling the maiden nanocraft sail. Besides, the CDR arrangement can sail continuously to distances close to infinity.

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